

8. MAZ'YA V.G. and PLAMENEVSKII B.A., On the coefficients in the asymptotic form of the solutions of elliptic boundary value problems in a region with conical points. *Math. Nachr.* 76, 1977.
9. MOROZOV N.F., *Collected Two-dimensional Problems of the Theory of Elasticity*. Leningrad, Izd-vo LGU, 1980.
10. KATO T., *Perturbation Theory for Linear Operators*. Berlin, Springer Verlag, 1980.
11. IVANOV L.A., KOTKO L.A. and KREIN S.G., Boundary value problems of variable regions. *Differential equations and their applications*. Coll. of papers. Vil'nyus, Izd-yo In-ta matematiki i kibernetiki AS LitSSR, 19, 1977.
12. MAZ'YA V.G., NAZAROV S.A. and PLAMENEVSKII B.A., *Asymptotic Form of the Solutions of Elliptic Boundary Value Problems when the Regions are Singularly Perturbed*. Tbilisi, Izd-vo Tbil. un-ta, 1981.
13. SI G. and LIBOVITS G., *Mathematic theory of brittle fracture*. In: *Fracture*. Moscow, Mir, 1975.
14. GRIFFITH A.A., *The theory of rupture*. In: *Proc. 1-st Intern. Congress for Appl. Mech.* Delft: Waltman, 1925.
15. MOROZOV N.F. and NAZAROV S.A., On the problem of computing the energy change in the Griffith problem. *Studies in Elasticity and Plasticity*. Coll. of papers. Leningrad, (zhd-vo LGU, 14, 1982.

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## STUDY OF CRACK OPENING USING THE WEIGHTING FUNCTIONS METHOD\*

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Some results of calculations of the opening of rectilinear, disc-like cracks under the action of a given system of forces, are given in /1-3/. A study of the opening of internal and surface cracks of more complex form is of interest, since in a number of cases it enables one to determine the depth of the crack from its known opening at the surface.

Formulas are obtained for the opening of elliptical, internal or surface cracks which occur when the body is acted upon by an arbitrary static load symmetrical about the plane of the crack.

1. Let us consider an elastic body with a rectilinear skew crack  $0 \leq x \leq l$ , internal or emerging at the surface  $x = 0$ . A weighting functions (WF) method was proposed in /4/ for computing the stress intensity factors (SIF) at the crack tip, and the possibility of using the method to determine the displacement field was suggested. When the elastic deformation energy  $W(l)$  and the displacement of the upper edge of the crack  $v(x, l)$  are both known for a certain external load, the WF can be found using the formula /4/

$$h(x, l) = \frac{1}{2} \left( \frac{1}{E'} \frac{\partial W}{\partial l} \right)^{-1/2} \frac{\partial v(x, l)}{\partial l} \quad (1.1)$$

where  $E' = E/(1 - \nu^2)$  for plane deformation,  $E' = E$  for the state of plane stress,  $E$  is the modulus of elasticity,  $\nu$  is Poisson's ratio and  $h(x, l)$  is independent of the type of loading.

We have the following formula for the SIF  $K(l)$  at the tip  $x = l$ :

$$K(l) = 2 \int_0^l \sigma(x) h(x, l) dx \quad (1.2)$$

where  $\sigma(x)$  denotes the arbitrary distribution of stresses applied to the crack edges. We also know that

$$K^2(l) = E' \partial W / \partial l \quad (1.3)$$

This enables us to rewrite (1.1) in the form

$$h(x, l) = 1/2 (E'/K(l)) \partial v(x, l) / \partial l \quad (1.4)$$

When the crack is internal, we have  $v(0, l) = 0$ , and this leads, according to (1.4), to  $h(0, l) = 0$ .

Thus, as was shown in /4/, when  $K(l)$  and  $v(x, l)$  are known, then no matter what the external load, formulas (1.2) and (1.4) will enable us to determine  $K(l)$  for any function  $\sigma(x)$ .

Let us now turn our attention to the problem of determining the displacement of the crack edge, and obtain  $v(x, l)$  from (1.4), taking into account the fact that  $v(l, l) = 0$ . We will have

$$v(x, l) = \frac{2}{E'} \int_x^l K(t) h(x, t) dt \quad (1.5)$$

Formulas (1.2) and (1.5) enable us to determine the opening of a crack when  $\sigma(x)$  is arbitrary, provided that the WF  $h(x, l)$  is known.

We note the following. Let us assume that concentrated unit forces are applied symmetrically to the upper and lower edge of the crack, at a distance  $\xi$  from the origin of coordinates, i.e.  $\sigma(x) = \delta(x - \xi)$ , where  $\delta(x - \xi)$  is the Dirac function. Denoting by  $K_0(\xi, l)$  the SIF at the tip  $x = l$  corresponding to the concentrated unit forces, we obtain from (1.2) the relation  $K_0(\xi, l) = 2h(\xi, l)$  whose substitution into (1.5) yields the formula

$$v(x, l) = \frac{1}{E'} \int_x^l K(t) K_0(x, t) dt \quad (1.6)$$

obtained earlier by Paris /1/ by a different method. Formulas (1.2) and (1.5) represent a specific example of formula (1.6) in the case when  $K(l)$  and  $K_0(\xi, l)$  are determined using the weighting functions method. However, the derivation of (1.5) using the above method is of independent interest, as it can be generalized to the case of a crack of more complex form.

Substituting (1.2) into (1.5) we obtain

$$v(x, l) = \frac{4}{E'} \int_x^l \left[ h(x, t) \int_0^t \sigma(\xi) h(\xi, t) d\xi \right] dt = \frac{4}{E'} \left\{ \int_0^x \left[ \sigma(\xi) \int_x^l h(x, t) h(\xi, t) dt \right] d\xi + \int_x^l \left[ \sigma(\xi) \int_\xi^l h(x, t) h(\xi, t) dt \right] d\xi \right\} \quad (1.7)$$

and this yields

$$v(x, l) = \frac{4}{E'} \int_0^l \sigma(\xi) f(x, \xi, l) d\xi, \quad f(x, \xi, l) = \int_{\max(x, \xi)}^l h(x, t) h(\xi, t) dt \quad (1.8)$$

Let us apply formulas (1.8) to the problem of a plane with a rectilinear skew crack. The WF given in /4/ for the crack tip  $h = l$  has the form

$$h(x, l) = \frac{1}{\sqrt{2\pi}} \sqrt{\frac{x}{l(l-x)}}$$

Calculations using the second formula of (1.8) yield

$$f(x, \xi, l) = \frac{1}{2\pi} \ln \left| \frac{\sqrt{x(l-\xi)} + \sqrt{\xi(l-x)}}{\sqrt{x(l-\xi)} - \sqrt{\xi(l-x)}} \right|$$

and the first formula of (1.8) leads to the expression for  $v(x, l)$  already obtained in /2/.

2. Let us consider a strip with a single notch  $0 \leq x \leq l$  emerging at the surface  $x = 0$ . In this case the WF has the form ( $H$  is the width of the strip) /5/

$$h(x, l) = \frac{1}{\sqrt{2\pi}} \frac{1}{\sqrt{l-x}} \left[ 1 + m_1 \frac{l-x}{l} + m_2 \frac{(l-x)^2}{l^2} \right]$$

$$m_j = A_j + B_j \lambda^2 + C_j \lambda^4, \quad j = 1, 2; \quad 0 \leq \lambda \leq 0.5; \quad \lambda = l/H$$

$$A_1 = 0.6147, \quad B_1 = 17.1844, \quad C_1 = 8.7822$$

$$A_2 = 0.2502, \quad B_2 = 3.2889, \quad C_2 = 70.0444$$

Let us specify the stress at the crack edges in the form of a polynomial in  $x$  ( $D_k$  are known function of  $l$ )

$$\sigma(x) = \sum_{k=0}^n D_k \left( \frac{l-x}{l} \right)^k$$

Then we obtain from (1.7) the displacement of the edge which at the strip surface when  $x = 0$ , takes the simple form

$$v(0, l) = \frac{4H}{\pi E'} \int_0^l (1 + m_1 + m_2) \left[ \sum_{k=0}^n D_k \left( \frac{1}{2k+1} + \frac{m_1}{2k+3} + \frac{m_2}{2k+5} \right) \right] d\lambda \quad (2.1)$$

The values of  $v(0, l)$  for the extension and flexure of the strip are given in /1/. Assuming that  $D_0 = \sigma_0$ ,  $D_k = 0$  when  $k > 0$  (extension) and  $D_0 = \sigma_0(1 - 2\lambda)$ ,  $D_1 = 2\sigma_0\lambda$ ,  $D_k = 0$  when  $k > 1$  (flexure), we obtain from (2.1) the values for  $v(0, l)$ , which are identical with those given in /1/.

3. Let us consider a disc-like crack of radius  $a$  in a body acted upon by an axisymmetric load. Repeating the arguments of Sect.1 we obtain for the WF, SIF and the displacement of the crack edge

$$\begin{aligned} h(r, a) &= \frac{1}{2\pi\sqrt{\pi}} \frac{1}{\sqrt{a(a^2 - r^2)}}, \quad K(a) = \frac{2}{\sqrt{\pi a}} \int_0^a \frac{\sigma(r) r dr}{\sqrt{a^2 - r^2}} \\ v(r, a) &= \frac{4}{\pi E'} \int_r^a \left[ \frac{1}{\sqrt{t^2 - r^2}} \int_0^t \frac{\sigma(\xi) \xi d\xi}{\sqrt{t^2 - \xi^2}} \right] dt = \\ &= \frac{4}{\pi E'} \left\{ \int_0^r \left[ \sigma(\xi) \xi \int_{\xi}^a \frac{dt}{\Delta} \right] d\xi + \int_r^a \left[ \sigma(\xi) \xi \int_{\xi}^a \frac{dt}{\Delta} \right] d\xi \right\} \\ (\Delta &= [t^2 - r^2]^{1/2} [t^2 - \xi^2]^{1/2}) \end{aligned}$$

Note that the last formula can be transformed to that given in /2/, by making the change of variable  $t = \xi/\sin \alpha$  in the inner integrals and changing the order of integration. Using elliptic integrals of the first kind, we reduce  $v(r, a)$  to the form

$$\begin{aligned} v(r, a) &= \frac{4}{\pi E'} \int_0^a \sigma(\xi) \frac{\xi}{M} \left[ F\left(\frac{m}{M}, \frac{\pi}{2}\right) - F\left(\frac{m}{M}, \arcsin \frac{M}{a}\right) \right] d\xi \\ m &= \min(\xi, r), \quad M = \max(\xi, r) \end{aligned}$$

In particular, for the centre of the crack edge we have

$$v(0, a) = \frac{4}{\pi E'} \int_0^a \sigma(\xi) \arccos \frac{\xi}{a} d\xi \quad (3.1)$$

Using the results in /3/, we can show that formula (3.1) can be generalized to the case of a non-axisymmetric load, i.e.  $\sigma = \sigma(r, \theta)$ . Then the function  $\sigma(\xi)$  in (3.1) will be replaced by

$$\sigma_0(\xi) = \frac{1}{2\pi} \int_0^{2\pi} \sigma(\xi, \theta) d\theta$$

4. Let us now consider the case of an internal crack bounded by an ellipse  $L$  with centre at the origin of coordinates, and the semi-axes  $a_x, a_y$  ( $a_x < a_y$ ). The stresses  $\sigma_z(x, y)$  are applied at the crack edges. We assume that  $\sigma_z(x, y)$  is an even function of  $x$  and  $y$ . Following /6/, we utilize the RMS values of the SIF at the crack edge, assuming that

$$\bar{K}_{x,y}^2 = \frac{1}{\delta A_{x,y}} K^2 dA, \quad \delta A_{x,y} = \pi a_{y,x} \delta a_{x,y}$$

with  $\bar{K}_x$  and  $\bar{K}_y$  depending on the variation in the parameters  $a_x$  and  $a_y$  respectively.

The WF corresponding to the parameters  $a_x$  and  $a_y$ , are determined from the formulas /6/

$$h_{x,y}(x, y, a_x, a_y) = \frac{1}{2} \left( \frac{\pi a_{y,x}}{E'} \frac{\partial W}{\partial a_{x,y}} \right)^{-1/2} \frac{\partial v(x, y, a_x, a_y)}{\partial a_{x,y}} \quad (4.1)$$

and are independent of the load. For  $\bar{K}_x, \bar{K}_y$  we have the formulas (A is the surface of the ellipse)

$$\bar{K}_{x,y}(a_x, a_y) = 2 \iint_A h_{x,y}(\xi, \eta, a_x, a_y) \sigma_z(\xi, \eta) dA \quad (4.2)$$

$$\bar{K}_{x,y}^2 = \frac{E'}{\pi a_{y,x}} \frac{\partial W}{\partial a_{x,y}} \quad (4.3)$$

Using relations (4.1) and (4.3), we shall express  $dv/\partial a_x$  and  $\partial v/\partial a_y$  in terms of  $\bar{K}_x$ ,  $\bar{K}_y$ ,  $h_x$  and  $h_y$ . After this we shall integrate the resulting expressions in  $a_x$  and  $a_y$ , respectively, requiring that the condition  $v|_L = 0$  holds. This will yield

$$v(x, y, a_x, a_y) = \frac{2\pi a_y}{E'} \int_{b_y(x,y)}^{a_x} \bar{K}_x(t, a_y) h_x(x, y, t, a_y) dt \quad (4.4)$$

or

$$v(x, y, a_x, a_y) = \frac{2\pi a_x}{E'} \int_{b_x(y,x)}^{a_y} \bar{K}_y(a_x, t) h_y(x, y, a_x, t) dt \quad (4.5)$$

$$b_{x,y}(x, y) = |x| [1 - (y/a_x, y)^2]^{-1/2}$$

where  $\bar{K}_x$ ,  $\bar{K}_y$  are obtained from formulas (4.2).

The WF  $h_x$  and  $h_y$  were obtained in /6/ and transformed to a form suitable for calculation in /7/ (a misprinted expression for  $h_y$  in /7/ should be corrected, namely, in the last term the numerator and denominator should be interchanged).

Expressions (4.4) and (4.5) yield the same quantity  $v(x, y, a_x, a_y)$ . However, all the above formulas were obtained on the assumption that  $a_x < a_y$ , therefore the lower limit on the right-hand side of (4.5) must satisfy the condition  $a_x < b_x(y, x)$ , or  $x^2 + y^2 > a_x^2$ , i.e. formula (4.5) can be used provided that the point  $x, y$  lies outside a circle of radius  $a_x$ .

5. Let us consider a semi-elliptical surface crack whose plane is perpendicular to the surface of the plate of thickness  $H$ . We shall assume that the centre of the ellipse is not displaced by the load and the crack contour remains semi-elliptical, i.e. only the magnitude of the semi-axes  $a_x$  and  $a_y$  change. The values  $\bar{K}_x^*$  and  $h_x^*$  corresponding to the parameter  $a_x$ , are connected by the relation ( $A^*$  is the surface of the semi-ellipse)

$$\bar{K}_x^*(a_x, a_y) = 4 \iint_{A^*} h_x^*(\xi, \eta, a_x, a_y) \sigma_z(\xi, \eta) dA \quad (5.1)$$

The WF  $h_x^*$  is obtained by multiplying  $h_x$  by the correction function /6, 7/. We must remember here that the correction function used in the present paper differs from that used in /7/ by a factor of 2.

Let us write the formula for the opening of the crack at its centre  $\delta = 2v(0, 0, a_x, a_y)$ , obtained from formula (4.4) taking into account the expression for the correction function  $f_n$  /6, 7/

$$\delta = \frac{24}{\pi E' a_y} \int_0^{a_x} \left\{ \frac{D(k) f_n(0, t/H)}{t E'^2(k) [E(k) + D(k)]} \times \right. \quad (5.2)$$

$$\left. \int_0^{a_y} d\eta \int_0^{t \zeta(\eta)} \left[ D(k) + \frac{E(k) \xi^2}{t \alpha} \right] \sqrt{\alpha} f_n(x_s, r) \sigma_z(\xi, \eta) d\xi \right\} dt$$

$$D(k) = E(k) - [F(k) - E(k)] \frac{t^2}{k^2 a_y^2}, \quad k^2 = 1 - \left( \frac{t}{a_y} \right)^2$$

$$\alpha = \zeta^2(\eta) - \left( \frac{\xi}{t} \right)^2, \quad x_s = \frac{\xi}{t \zeta(\eta)}, \quad r = \frac{t \zeta(\eta)}{H}$$

$$\zeta(\eta) = \left[ 1 - \left( \frac{\eta}{a_y} \right)^2 \right]^{1/2}$$

The opening of the crack was calculated using formula (5.2), for the cases of extension and flexure of the plate. The results obtained are compared with the opening  $\delta^*$  of a crack inclined to the  $y$  axis, calculated from the data in /1/ (or from formula (2.1)). Fig.1 shows the dependence of  $\lambda = a_x/(2a_y)$  on the parameter  $\kappa = \delta^*/\delta$  characterizing the effect of the form of the crack on its opening when the depth  $a_x$  is given, for  $H/a_x = 3$  (the dashed line) and  $H/a_x = 100$  (the solid line), for two types of stress state (1 denotes extension and 2 denotes flexure). Using Fig.1 we can determine the opening of a semi-elliptical crack with parameters  $a_x$  and  $a_y$  by dividing the opening of a skew crack of depth  $a_x$  by the corresponding form parameter  $\kappa$ . We note that the relative size of the crack and the plate thickness substantially affect the magnitude of  $\kappa$ .

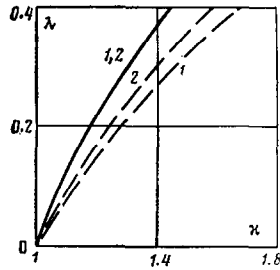


Fig.1

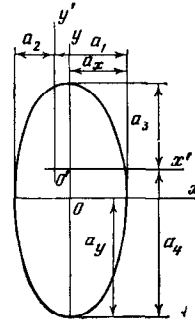


Fig.2

6. We derived in Sect.4 formulas for the opening of an elliptical crack in the case when  $\sigma_z(x, y)$  is an even function, i.e. in the case when the centre of the ellipse is not displaced when a load is applied. If this condition does not hold, then following /6/ it is necessary to consider a crack characterized by four parameters  $a_1, a_2, a_3, a_4$  (Fig.2).

The WF corresponding to the parameters  $a_1, a_2$ , have the form

$$h_{1,2} = h_x \pm \left[ \frac{\partial a_x}{E(k) + D(k)} \right]^{1/2} \frac{x}{\pi a_x a_y \sqrt{\alpha}}$$

$$a_x = \frac{a_1 + a_2}{2}, \quad a_y = \frac{a_3 + a_4}{2}, \quad x = x' - \frac{a_1 - a_2}{2}, \quad y = y' - \frac{a_3 - a_4}{2}$$

The SIF are written thus

$$\bar{K}_{1,2} = 2 \iint_A h_{1,2}(x', y', a_1, a_2, a_3, a_4) \sigma_z(x', y') dA$$

and the displacement of the crack edge is

$$v = \frac{2\pi a_y}{E'} \int_{b_1^+}^{a_1} \bar{K}_1(t, a_2, a_3, a_4) h_1(x', y', t, a_3, a_4) dt, \quad x' \geq \frac{a_1 - a_2}{2}$$

$$v = -\frac{2\pi a_x}{E'} \int_{-a_4}^{b_2^-} \bar{K}_2(a_1, t, a_3, a_4) h_2(x', y', a_1, t, a_3, a_4) dt, \quad x' \leq \frac{a_1 - a_2}{2}$$

$$b_j \pm = [2x' \pm a_j (1 - \zeta(y))] (1 + \zeta(y))^{-1}, \quad j = 1, 2$$

It is clear that the quantity  $v$  found using the WF  $h_3, h_4$  corresponding to the parameters  $a_3, a_4$  is identical with the one given above.

The numerical work was carried out by E.A. Berezina.

#### REFERENCES

1. TADA H., PARIS P.C. and IRWIN G.R., The stress analysis of crack handbook, Heller-town: Providence, Del Res. Corp., 1973.
2. PANASYUK V.V., Critical Equilibrium of Brittle Bodies with Cracks. Kiev, Naukova Dumka, 1968.
3. SLEPYAN L.I., Mechanics of Cracks. Leningrad, Sudostroyeniye, 1981.
4. RICE J.R., Some remarks on elastic crack-tip stress fields. Intern. J. Solids and Struct., 8, 6, 1972.
5. BUECKNER H.F., Weight functions for the notched bar. Z. angew. Math. und Mech., 51, 2, 1971.
6. BESUNER P.M., The influence function method for fracture mechanics and residual fatigue life analysis of cracked components under complex stress fields. Nucl. Engng. and Design. 43, 1, 1977.
7. NOVOZHILOV V.V. and RYBAKINA O.G., Study of the propagation of surface cracks under cyclic loads. In Mechanical Metal Fatigue. Kiev, Nakova Dumka, 1983.